

1 excitation [3, 4]. E.g., vibration measurements, taken during the operation stage of a concrete bridge or a
2 prestressed steel wind tower, are useful applications to control their calculation model, evaluate their
3 stiffness or monitor their internal stresses [5, 6]. Bonopera et al. [7] fully illustrated a state-of-the-art
4 review of researches conducted worldwide about this topic. Particularly, Hamed and Frostig [8], Jaiswal
5 [9], Limongelli et al. [10] and Bonopera et al. [11, 12] declared that the natural frequencies and stiffness
6 of Prestressed Concrete (PC) girder-bridges, with a parabolic or a straight tendon, only significantly vary
7 under the effect of crack initiation or crack re-opening. Pisani et al. [13] confirmed that, for prestressed
8 beams, modal parameters provide information regarding damage only when the members are close to the
9 ultimate conditions. Previously, Noble et al. [14, 15] executed several experiments on a number of post-
10 tensioned steel and concrete beams in small-scale. The researchers declared that the small reduction in
11 fundamental frequency with increasing post-tensioning force is not related to the softening effect. Thus,
12 they deduced that the compression-softening theory must be eliminated from discussion of all forms of
13 pre and post-tensioned structures. The researchers also claimed that the dynamic effect of a compressive
14 force and that of a post-tensioning on a beam are different on a phenomenological level. Consequently,
15 they retained that a prestressing force should always be considered as an axial force internally applied.
16 Gan et al. [2] sustained that this divergence in Noble et al. [15] was caused by the closure effect of
17 shrinkage cracks and/or microcracks inside the post-tensioned concrete beams which, in turn, was not
18 considered in the theoretical predictions of fundamental frequency. According to the literature above
19 mentioned, there is a meaningful disagreement among researchers about the effect of pre and post-
20 tensioning forces on the beam dynamics. Therefore, it was deemed that the results regarding this topic
21 need additional experimental investigations on steel beams rather than those in reinforced concrete, with
22 the aim to prevent stiffening effects caused by the microcrack closure [9] and by the time-increment of
23 Young's modulus regarding the consolidation/hardening of concrete [11, 12, 16]. Such investigations had
24 to be conducted on post-tensioned beams in a larger scale for avoiding that any mass at the end constraints,
25 usually composed of load cells and/or hydraulic jacks, was greater than the self-mass of the beams
26 themselves. In fact, it is what occurred in Noble et al. [14] where the fundamental frequencies were
27 affected by such masses [11, 12]. Accordingly, the conclusions of Noble et al. [14] required to be revised.

1 Hence, this work was mainly necessary for properly clarifying and evaluating the dynamics of prestressed
2 beams. Indeed, a correct evaluation of fundamental frequency is crucial for designing new bridges and
3 prestressed wind turbine towers, monitoring and determining the conditions of existing ones and related
4 components. Furthermore, the reproduction of experimental data using correct algorithms enables
5 effective interpretations of in situ measurements and simulations of virtual laboratory tests [17].

6 This work was a research sub-program developed at the National Center for Research on
7 Earthquake Engineering (NCREE) according to a campaign study on girder-bridges which initiated in
8 2015 by testing PC beam specimens [11, 12, 16–20]. Given the conflicts in the aforementioned findings,
9 firstly, free transverse vibrations were induced to a simply supported post-tensioned thin-walled steel-
10 box-girder in larger scale and with eccentric straight cables. Different post-tensioning forces were applied.
11 The specimen was characterized by initial second-order curvatures, due to the post-tensioning force
12 causing two equal bending moments at the end constraints, and an axial end constraint, due to the stiffness
13 of the cables [21, 22]. Secondly, the experiments executed by Noble et al. [14] were analyzed. Indeed, the
14 researchers similarly investigated the compression-softening theory in two simply supported thin-walled
15 steel-box-members post-tensioned by a straight cable. The two specimens in small scale had different
16 slenderness. Subsequently, the fundamental frequencies measured from testing, and those obtained by
17 Noble et al. [14] 's experiments, were compared with a formula based on the Timoshenko shear beam
18 model [23], where the post-tensioning force is equivalent to a compressive force externally applied. The
19 frequencies were also compared with two high-fidelity Finite-Element (FE) models including shear
20 deformation, in which the post-tensioning force is applied as initial tension in the cables. In the FE
21 analyses, the contact between cables and surrounding sections was additionally simulated. According to
22 the obtained findings, the dynamic effect of a compressive force and that of a post-tensioning are
23 phenomenologically different only when the cables are in contact with the surrounding beam's section.
24 Depending on this feature, the dynamics of all forms of pre and post-tensioned beams is ruled by
25 compression-softening effect. Similarly to what deduced for PC girder-bridges [7, 19, 20], the
26 fundamental frequency as parameter for post-tensioning loss identification is made doubtful in thin-
27 walled steel-girder-bridges draped with a deviator.

1 **2. Reference solutions**

2 The reference solution, proposed in this study, describes the prestressed beam dynamics based on
 3 Timoshenko beam model, where transverse shear deformability is considered [23, 24]. The shear beam
 4 theory is often used to model the behavior of structures both for stability or dynamic analyses [25–27].
 5 The effect of rotational inertia is neglected because it may be appreciable only at high frequencies [28].
 6 Particularly, the vibrational response of an axially unloaded simply supported beam takes first-order
 7 effects into account including shear deformation [29]. Vice versa, an externally compressed member of
 8 length L allows to consider the second-order effects in a beam pre or post-tensioned by a straight tendon
 9 which can be or not in contact with the surrounding section (Fig. 1). This configuration characterized the
 10 experimental investigations illustrated in Section 3.1. Indeed, a thin-walled steel-box-girder was subjected
 11 to different post-tensioning forces ($N_{0x,aver}$) exerted by eccentric straight cables which, in turn, were free
 12 to vibrate inside the Rectangular Hollow Section (RHS) [30] (Sections 3.2 and 3.3). In short, the cables
 13 were only in contact at the girder ends. Their cross sectional area were assumed as unchanged after
 14 deformations. Therefore, the eccentricity (e) of post-tensioning forces ($N_{0x,aver}$) was theoretically zero.
 15 The flexural shape $v^{(0)}$ of the girder’s fundamental frequency ($f_{shear,1,II}$), with total self-mass per unit length
 16 m_{tot} , is represented in Fig. 1, according to the shear beam model which, in turn, included the second-order
 17 effects.

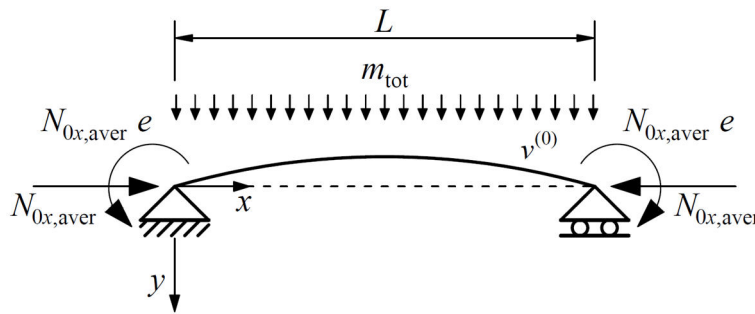


Fig. 1. Reference solution. Prismatic beam pre or post-tensioned by eccentric straight tendon. Fundamental flexural shape $v^{(0)}$ according to the shear model. The dashed line represents the undeformed configuration.

18 When the member in Fig. 1 meets all the requirements of Euler-Bernoulli theory, it is subjected to small
 19 displacements and rotations. Thus, the second-order n th natural frequency ($f_{E.-B.,n,II}$) of a simply supported
 20 post-tensioned steel-girder with a straight unbonded tendon is described by the following equation where,
 21 the effects of longitudinal vibrations and warping are not considered [23, 29]:
 22

$$f_{E-B,n,II} = \frac{1}{2\pi} \sqrt{\frac{\alpha^2 n^4 \pi^4}{L^4} \left(1 - \frac{N_{0x,aver}}{N_{crE,n}}\right)} \quad n = 1, 2, \dots \quad (1)$$

1
2 Here the coefficient $\alpha^2 = EI g / m_{tot}$ contains Young's modulus E and cross sectional second moment of the
3 area I of the thin-walled box-girder. The gravitational acceleration $g = 9.81 \text{ m/s}^2$. Instead, the second
4 square root contains the n th critical buckling load of the simply supported girder given by $N_{crE,n} = n^2 \pi^2 EI$
5 $/L^2$. The self-mass per unit length of the straight unbonded tendon (m_{tendon}) does not affect the beam
6 dynamics. Likewise, no stiffening effect is induced by the cables under post-tensioning ($N_{0x,aver}$).
7 Assuming $n = 1$ in Eq. (1), the second-order fundamental frequency ($f_{E-B.,1,II}$) is obtained. In detail, Eq.
8 (1) reduces to Eq. (2) reported in Bonopera et al. [11], where the n th critical buckling load ($N_{crE,n}$) becomes
9 equal to the Euler buckling load $N_{crE,1} = \pi^2 EI / L^2$. Furthermore, disregarding the term containing the post-
10 tensioning force $N_{0x,aver}$, Eq. (3), likewise reported in Bonopera et al. [11], is achieved. Such formula
11 represents the first-order fundamental frequency ($f_{E-B.,1,I}$) of simply supported PC girder-bridges with a
12 straight bonded or unbonded tendon. Indeed, the dynamic effect of a pre or post-tensioning force ($N_{0x,aver}$)
13 is negligible in these types of PC girder-bridges [11]. Conversely, when the aforementioned thin-walled
14 girder meets all the requirements of the shear beam model, its first-order n th natural frequency ($f_{shear,n,I}$) is
15 expressed by the following equation which, in turn, does not consider longitudinal vibrations, warping
16 and rotary inertia [29]:

$$f_{shear,n,I} = \frac{1}{2\pi} \sqrt{\frac{\alpha^2 n^4 \pi^4}{L^4 \left(1 + \frac{n^2 \pi^2 r^2}{L^2} \frac{E}{k G}\right)}} \quad n = 1, 2, \dots \quad (2)$$

17
18 In Eq. (2), the coefficient $r^2 = I/A$, where A is the cross sectional area of the thin-walled girder. Elastic
19 shear modulus $G = E / [2(1 + \nu)]$. ν is Poisson's ratio. k is instead the shear coefficient for generic thin-
20 walled hollow sections which is described by the following equation [31]:

$$k = \frac{10(1+\nu)(1+3m)^2}{(12+72m+150m^2+90m^3) + \nu(11+66m+135m^2+90m^3) + 10n^2[(3+\nu)m+3m^2]} \quad (3)$$

21
22 Here we have the parameters $m = b t_1 / h t$ and $n = b / h$, where internal width b , thickness of the flanges t_1 ,
23 height h , and thickness of the web t . According to the above, based on the shear model, and when second-
24 order effects are assumed, the n th natural frequency ($f_{shear,n,II}$) of a simply supported post-tensioned thin-
25 walled steel-girder with a straight unbonded tendon (Fig. 1) can be expressed as:

$$f_{\text{shear},n,II} = \frac{1}{2\pi} \sqrt{\frac{\alpha^2 n^4 \pi^4}{L^4 \left(1 + \frac{n^2 \pi^2 r^2}{L^2} \frac{E}{kG}\right)}} \sqrt{1 - \frac{N_{0x,aver}}{N_{\text{crE,shear},n}}}. \quad n = 1, 2, \dots \quad (4)$$

1
2 In Eq. (4), the second square root, originating from Eq. (1), contains the n th critical buckling load with
3 the effect of shear deformation ($N_{\text{crE,shear},n}$) which is given by the following approximate formula [23]:

$$N_{\text{crE,shear},n} = \frac{N_{\text{crE},n}}{1 + (N_{\text{crE},n}/GA_0)}. \quad n = 1, 2, \dots \quad (5)$$

4
5 In Eq. (5), the n th critical buckling load of a simply supported girder $N_{\text{crE},n}$, whereas the parameter $A_0 = A$
6 $/m$. m is the correction coefficient which takes into account the nonuniform distribution of the shear
7 stresses throughout the cross sectional area (A). For a generic thin-walled section, which bends in the
8 plane of the web, $m \simeq A/A_w$. A_w is the cross sectional area of the web. Substituting $N_{\text{crE},n} = n^2 \pi^2 EI/L^2$ for
9 the denominator, and the expression $I = r^2 A$ then, Eq. (5) can be reformulated by Eq. (6) which is reported
10 as follows [23]:

$$N_{\text{crE,shear},n} = \frac{N_{\text{crE},n}}{1 + n^2 m \left[\pi/(L/r)\right]^2 (E/G)}. \quad n = 1, 2, \dots \quad (6)$$

11
12 We see that the smallest critical buckling load ($N_{\text{crE,shear},n}$) occurs for $n = 1$. When slenderness (L/r) is
13 sufficiently large, shear effects are negligible, and the classical Euler solution is recovered. From Eq. (6)
14 we observe that shear strains decrease the critical load ($N_{\text{crE,shear},n}$). Therefore, the shear correction
15 becomes significant for short post-tensioned thin-walled girders (small L/r) made with a high strength
16 steel, so that the short beam still fails due to buckling rather than yield [23].

17 18 **3. Experimental investigation**

19 **3.1. Thin-walled steel-box-girder post-tensioned by eccentric straight cables**

20 Given the discrepancies created about the effect of pre and post-tensioning forces on the beam dynamics
21 (Section 1), a thin-walled RHS member of $b = 150$ mm, $h = 350$ mm, $t = t_1 = 9$ mm, and made with a high
22 strength steel, was adopted (Fig. 2). Steel was chosen to prevent any stiffening effects caused by the crack
23 and/or microcrack closure [2, 9] and by the time-increment of Young's modulus regarding the
24 consolidation/hardening of concrete [11, 12, 16]. The member was subjected to a series of post-tensioning
25 forces exerted by a straight unbounded tendon with a small eccentricity ($e/h = 0.23$) with respect to the

1 RHS centroid (Fig. 2). The tendon was composed of 5 steel “seven–wire” strands, of 15.2 mm in diameter,
 2 which passed internally through the RHS body and only anchored at its ends (Fig. 2). The tendon’s self-
 3 mass per unit length ($m_{\text{tendon}} = \rho_{\text{tendon}} \times A_{\text{tendon}}$) was 0.0533 kN/m. Two steel supports were arranged at the
 4 RHS ends to create pinned-end restraints, resulting in a clear span L (Fig. 2). $132 \times 176 \times 6 \text{ mm}^3$ vertical
 5 stiffeners, made with the same structural steel, were welded into the RHS upper part at a distance of ≈ 1.0
 6 m to prevent instability modes with respect to the girder’s vertical axis under post-tensioning (Fig. 2).
 7 Furthermore, the RHS ends were covered by a $250 \times 450 \times 20 \text{ mm}^3$ steel plate, likewise made with the same
 8 steel. 12 threaded bars, of 16 mm in diameter, were utilized for the fastening, for a total mass at the RHS
 9 ends of 317.1 N [Figs. 3(a)–(b)]. All corresponding material and geometric properties of the thin-walled
 10 RHS member were listed in Table 1. Notably, unit weight, Young’s modulus and yielding stress of high
 11 strength steel were evaluated by tensile tests considering the average values obtained from 4 hourglass
 12 specimens [32].

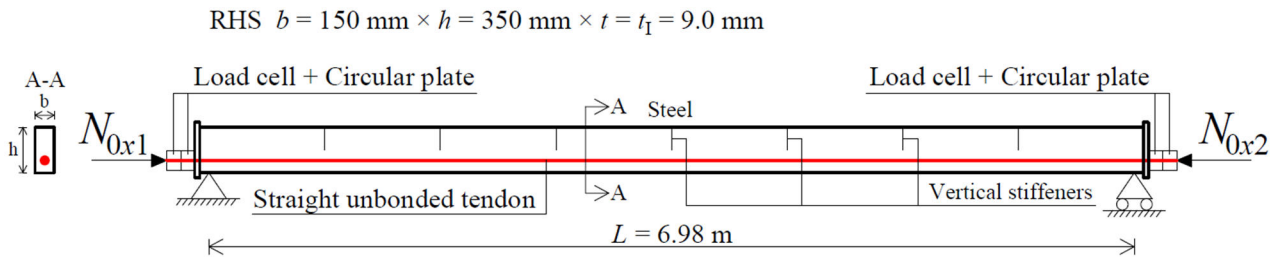


Fig. 2. Thin-walled steel RHS member post-tensioned by eccentric straight unbonded cables.

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Table 1. Properties of the thin-walled steel RHS member (Fig. 2).

Property	RHS $50 \times 30 \times 3 \text{ mm}^3$
$I \text{ (mm}^4\text{)}$	1.33399×10^8
$A \text{ (mm}^2\text{)}$	8,676
$A_w \text{ (mm}^2\text{)}$	5,976
Slenderness	56
$k \text{ [Eq. (3)]}$	0.687
$L \text{ (m)}$	6.98
$\rho_{\text{beam}} \text{ (kN/m}^3\text{)}$	77.01
$E \text{ (GPa)}$	205.12
$f_{yk} \text{ (MPa)}$	390
$G \text{ (GPa)}$	82.05
ν	0.25
$A_{\text{tendon}} \text{ (mm}^2\text{)}$	695
$\rho_{\text{tendon}} \text{ (kN/m}^3\text{)}$	76.65
$E_{\text{tendon}} \text{ (GPa)}$	200
$\sigma_{ty} \text{ (MPa)}$	1,860
$e \text{ (mm)}$	80

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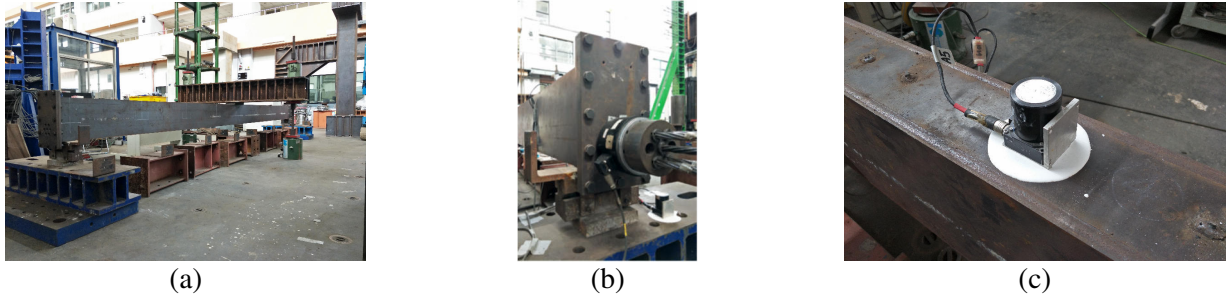


Fig. 3. (a) Laboratory test rig. (b) Load cell and circular plate at one RHS end. Reference seismometer (Af) to the floor (Fig. 4). (c) Seismometer A1 on the top of quarter cross section (Fig. 4).

1

2

3.2. Experimental set-up

3

The thin-walled steel-box-girder was located in a test rig [Fig. 3(a)]. At one end constraint, a hydraulic

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jack, of 1,000 kN in force capacity, was utilized to apply the post-tensioning forces by pulling the cables

5

outward (Fig. 2). Yet, a load cell, of 130 mm in diameter, 120 mm in length, 1,000 kN in force capacity,

6

2 mV/V in sensitivity, and 78.5 N in mass, was fixed at either RHS ends to measure the applied post-

7

tensioning forces N_{0x1} and N_{0x2} [Fig. 2 and Fig. 3(b)]. A circular steel plate, of 130 mm in diameter, 60

8

mm in length, and 49.4 N in mass, was additionally located between the load cells and the end of cables

9

[Fig. 3(b)]. 6 mean post-tensioning forces ($N_{0x,aver}$) were totally applied in values of $\approx 117, 214, 332, 421,$

10

501 and 598 kN, to respectively induce second-order effects of 2.2, 4.0, 6.4, 8.2, 9.9 and 12.1% of the

11

girder's critical buckling load $N_{crE,1}$ or $N_{crE,hear,1} = 5,543$ kN. Notably, the maximum tension force in each

12

cable was not to exceed the value of ≈ 140 kN, according to the laboratory's safety conditions. Thus, the

13

maximum tensile strength reached in the tendon, $\sigma_{tendon,max} = N_{0x,aver,max} / A_{tendon} = 598 \text{ kN} / 695 \text{ mm}^2 = 860$

14

MPa, was $\approx 46\%$ of yielding stress σ_{uy} (Table 1). The different post-tensioning forces (N_{0x1} and N_{0x2})

15

measured at the RHS ends occurred because of the post-tensioning losses due to anchorage slips and

16

relaxation of cables (Fig. 2). The measuring system included 5 seismometers installed along the RHS

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body (Af, A0, A1, A2 and A3) as depicted in Fig. 4. Particularly, 5 high-precision servo velocity

18

seismometers, lightweight (2.65 N), of 50 mm in diameter, 70 mm in length, and 5 mV/gal in sensitivity,

19

were selected (VSE-15D, Tokyo Sokushin). 2 seismometers of which, marked as A1 and A2, were

20

vertically fixed on the top of quarter and midspan cross sections, i.e., at $i = 1$ and 2, with the aim to

21

respectively collect acceleration data with respect to the RHS strong axis [Fig. 3(c) and Fig. 5(c)]. Vice

22

versa, the 2 seismometers, marked as A0 and A3, were located on the top of the RHS ends (Fig. 4). The

23

seismometer Af was instead placed as a reference to the floor, i.e., near the RHS end at $i = 0$ [Fig. 3(b)].

1 **A_f** was used to record eventual anomalies of the measuring system. All seismometers were connected to
 2 a signal conditioner and, later, to a data logger placed on a desk near the test rig. The experimental set-up
 3 in Fig. 4 shows their positions in violet. The thin-walled box-girder's self-mass per unit length (m_{beam}),
 4 comprehensive of 4 seismometers (**A0**, **A1**, **A2** and **A3**) and horizontal steel plates for positioning a
 5 number of displacement transducers, was 0.7167 kN/m. Specifically, such plates, of $300 \times 120 \times 7.5 \text{ mm}^3$
 6 in dimension, 200 mm in length, and made with the same structural steel of the member, were welded
 7 along the RHS body [Figs. 5(a)–(b) and Fig. 5(d)]. Notably, the measurements obtained by such
 8 transducers were objective of a different study.

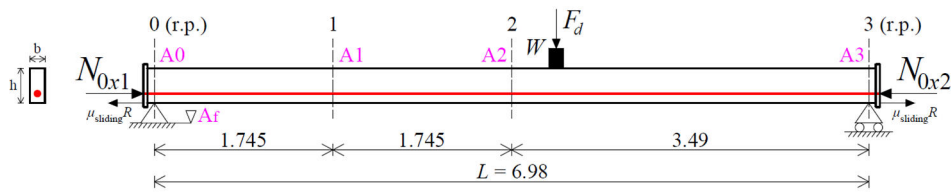


Fig. 4. The experimental set-up with the positions of hydraulic jack (W) and servo velocity seismometers (**A_f**, **A0**, **A1**, **A2** and **A3**). Units: m.

9



Fig. 5. (a) Hinge support. (b) Roller support. (c) Hydraulic jack (W) on a rebar anchored near the midspan and seismometer **A2** (Fig. 4). (d) Hydraulic hand pump to the floor before activation.

10

11 **3.3. Free transverse vibrations**

12 Free transverse vibrations were performed in a short term on the thin-walled box-girder following the
 13 application of $N_{0x,aver}$, in values of $\approx 117, 214, 332, 421, 501$ and 598 kN , pulling the cables outward
 14 (Section 3.2). The girder was thus subjected to rotational boundary conditions due to the bending moments
 15 $N_{0x,aver} e$ at the RHS ends (Fig. 4). All seismometers (**A_f**, **A0**, **A1**, **A2** and **A3**) acquired the acceleration
 16 data at a sampling rate of 200 Hz, and with a Block Size (BS) of 131,072 samples. Vibration
 17 measurements were repeated thrice after that each post-tensioning force ($N_{0x,aver}$) was assigned. 18
 18 experiments were totally executed. Particularly, free vibrations were induced pulling out a steel rebar, of
 19 6 mm in diameter, welded at a distance not greater than 600 mm from the midspan ($i = 2$) both at right

1 and left side (Fig. 4). Such procedure was feasible using a hydraulic oil jack W , of 120 mm in diameter,
2 170 mm in length, 100 kN in force capacity, and 118.5 N in mass, placed on the top of the girder and
3 crossed by each rebar [Fig. 5(c)]. These latter were connected to the jack (W) at their higher end.
4 Specifically, the jack pulled out every rebar from the girder counteracting its force on the girder itself.
5 The jack was in turn actuated by a hydraulic hand pump, of 96.53 MPa in pressure capacity, positioned
6 to the floor [Fig. 5(d)]. Therefore, the thin-walled box-girder with initial second-order curvature, due to
7 the bending moments ($N_{0x,aver} e$), total self-mass ($m_{tot,beam} = m_{beam} \times L = 5.0$ kN), and an axial end constraint,
8 due to the stiffness of the cables [21, 22], was vibrated by small unbalanced forces caused by the
9 unavoidable weld's breaking between rebar and member. In turn, weld's breaking was due to its lowest
10 resistance (Fig. 4). Consequently, the value of the release forces (F_d) were of difficult estimations (Fig. 4)
11 [17]. The vibrations were measured with respect to the RHS horizontal (strong) axis. The post-tensioning
12 forces N_{0x1} and N_{0x2} were measured every second for a total of 200 seconds by a data acquisition unit
13 using a different data logger. The average measurements of N_{0x1} and N_{0x2} for one repetition of vibration
14 tests were listed in Table 2. The cables were never in contact with the surrounding RHS during vibrations
15 (Fig. 4) unlike when the thin-walled box-girder was unloaded ($N_{0x,aver}=0$). In this specific case,
16 measurements were discarded. Notably, the eccentric post-tensioning force, when is assumed as
17 externally applied, causes the yielding stress ($f_{yk} = 390$ MPa) of the most extreme fibres of midspan cross
18 section at a value of $N_{0x,ULS,0.23} \approx 1,007$ kN (Fig. 4). $N_{0x,ULS,0.23}$ corresponds to $\approx 18.2\%$ of the critical
19 buckling load $N_{crE,1}$ or $N_{crE,hear,1} = 5,543$ kN.

20 21 **3.4. Vibration frequencies**

22 Figs. 6(a)–(b) show the acceleration from time-histories of the thin-walled box-girder, with corresponding
23 Fast Fourier Transform (FFT) results, when the post-tensioning force $N_{0x,aver} = 214$ kN (Fig. 4). BS =
24 65,536 samples. Particularly, Fig. 6(a) refers to the instrumented section (A1) at the quarter ($i = 1$), whilst
25 Fig. 6(b) refers to the instrumented section (A2) at the midspan ($i = 2$). The raw signals (acceleration-time
26 data) of each test repetition were imported into MATLAB [33] where a signal processing algorithm
27 eliminated significant electrical noise [16]. The peak picking method was then adopted. Fundamental and
28 second-mode frequencies were located at each peak of the FFT, as indicated in Figs. 6(a)–(b). Same

1 procedure was done assuming a BS = 131,072 samples. A total of 72 of which were gathered, since each
 2 free vibration was repeated thrice, i.e., when the girder was subjected to each $N_{0x,aver}$. Figs. 6(a)–(b)
 3 appertain both to one test repetition. Notably, the maximum $N_{0x,aver,max} = 598$ kN was $\approx 59\%$ of the
 4 maximum allowable $N_{0x,ULS,0.23} \approx 1,007$ kN (Section 3.3).

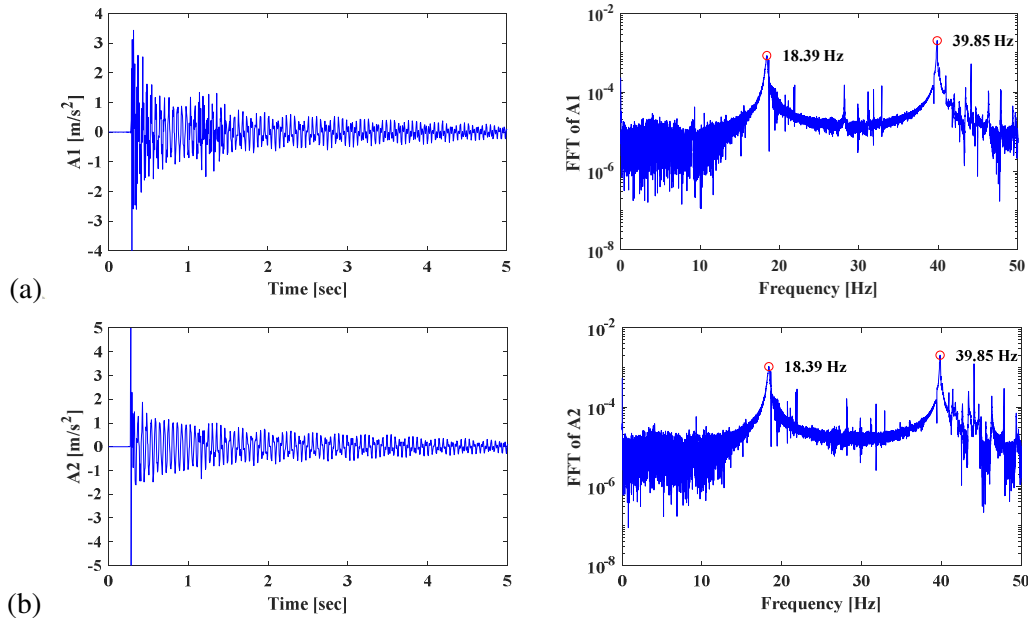


Fig. 6. Acceleration from time-histories and FFT results for $N_{0x,aver}=214$ kN and BS=65,536 samples. Seismometers: (a) A1; (b) A2 (Fig. 4).

5
 6 Seismometers A1 and A2 provided a slight increasing in fundamental frequency $f_{1,exp}$ (almost
 7 constant) from 18.37 to 18.39 Hz into the initial increment in post-tensioning force ($N_{0x,aver}$) which went
 8 from 117 to 214 kN. Subsequently, the girder was subjected to a significant softening effect: the frequency
 9 ($f_{1,exp}$) underwent a prominent decrement of 0.93 Hz under an increment in $N_{0x,aver}$ which went up to 598
 10 kN. A maximum $f_{1,exp,max} = 18.39$ Hz and a minimum frequency $f_{1,exp,min} = 17.46$ Hz were finally measured
 11 within the experimental tendency. All frequencies ($f_{1,exp}$) identified using seismometers A1 and A2, and a
 12 BS = 65,536 samples within the FFT of the 3 repetitions, were listed in Tables 2–3. The trend of $f_{1,exp}$ was
 13 confirmed assuming a BS = 131,072 samples, i.e., increasing the accuracy in frequency estimations (200
 14 Hz / 131,072 = 0.002 Hz). The only difference was the measurement $f_{1,exp} = 17.91$ Hz when the applied
 15 post-tensioning force $N_{0x,aver} = 332$ kN. Besides, seismometers A1 and A2 always furnished equal
 16 measurements ($f_{1,exp}$) in relation to the same configuration of free vibration test (Section 3.3). Notably, the
 17 masses composed of load cell, threaded bars, circular and rectangular steel plate at one RHS end [Fig.
 18 3(b)] were only $\approx 8.9\%$ of the girder's total self-mass ($m_{tot,beam} = 5.0$ kN). Thus, conversely to what

1 happened in the experiments of Noble et al. [14], it was avoided that such masses, lower than the total
 2 self-mass ($m_{\text{tot,beam}}$), affected the beam dynamics. Contrariwise, the second-mode frequencies were
 3 affected by the application of weak release forces (F_d) (Fig. 4). To obtain reliable second-mode frequency
 4 measurements through FFT, strong excitations are generally required to be induced during vibration
 5 testing [17]. Seismometer A1 was strategically located at a distance of 1.745 m from the midspan (at $i =$
 6 1) since the latter is a nodal point for the second-mode vibration of a beam.

7 8 **4. Interpretation of experimental results**

9 **4.1. Comparison with reference solutions**

10 The fundamental frequencies ($f_{1,\text{exp}}$) were firstly compared with Eqs. (1)–(2) and Eq. (4), in which, $n = 1$,
 11 whereas the corresponding critical buckling loads, $N_{\text{crE},1}$ and $N_{\text{crE, shear},1}$, were respectively assumed
 12 (Section 2). The comparisons were presented in Table 2, in which percentage errors were formulated by
 13 $\Delta = (f_1 - f_{1,\text{exp}}) / f_{1,\text{exp}}$. Specifically, $f_{\text{E.-B.,1,II}}$ decreased from ≈ 18.91 to 18.42 Hz into the increment in post-
 14 tensioning force ($N_{0x,\text{aver}}$) went from ≈ 300 to 598 kN [Eq. (1)] corresponding to an average error
 15 $\Delta_{\text{aver}}=5.5\%$. Vice versa, $f_{\text{shear},1,II}$ decreased from ≈ 18.80 to 18.32 Hz into the same range [Eq. (4)]
 16 corresponding to an average error $\Delta_{\text{aver}}=4.8\%$. The frequencies ($f_{1,\text{exp}}$) followed the significant decreasing
 17 tendency of Eq. (1) and Eq. (4) and concurrent with the compression-softening theory. Therefore, the
 18 trend of frequencies ($f_{1,\text{exp}}$) disagreed with findings and deductions of Noble et al. [14]. Yet, the differences
 19 between frequencies $f_{\text{E.-B.,1,II}}$ and $f_{\text{shear},1,II}$ of ≈ 0.10 Hz (Table 2) were caused by the effect of shear
 20 deformation in Eq. (4). By contrast, the slight initial increment in $f_{1,\text{exp}}$ of 0.02 Hz, i.e., into the range 117
 21 $\text{kN} \leq N_{0x,\text{aver}} \leq 214$ kN, was not of the same rate as predicted by Eq. (1) and Eq. (4). Such trend occurred
 22 under a lower amount of second-order effects ($\leq 4.0\%$ of $N_{\text{crE},1}$ or $N_{\text{crE, shear},1}$), i.e., without a considerable
 23 softening effect along the thin-walled girder. Notably, the total self-mass per unit length (m_{tot}) considered
 24 the mass of hydraulic jack, $W = 118.5$ N [Fig. 5(c)], which was distributed along the length (L) of the
 25 member [$m_{\text{tot}} = m_{\text{beam}} + (W/L) = 0.7337$ kN/m]. W was equal to 2.4% of the total self-mass ($m_{\text{tot,beam}}=5.0$
 26 kN). Based also on published works and previous results [12, 17, 34, 35], masses of sensors or devices
 27 (affixed transversally) affect the fundamental frequency ($f_{1,\text{exp}}$) of a simply supported beam when are
 28 greater than $\approx 0.3\%$ of its total self-mass. This consideration is valid when strong excitations are induced

1 by an impact or release force (F_d) because the precision in detection of frequency ($f_{1,\text{exp}}$) is related to the
 2 complexity of signal's response, and to the proportion of vibrational response assigned to the fundamental
 3 mode [14, 15, 17]. FE analyses in Strand7 [36], discretizing the thin-walled box-girder into 20 elements,
 4 and accounting for total self-mass per unit length (m_{tot}), girder's axial deformations and curvatures,
 5 depending on post-tensioning ($N_{0x,\text{aver}}$) and bending moments ($N_{0x,\text{aver}} e$) at the RHS ends (Fig. 4),
 6 furnished the same frequencies $f_{\text{E.-B.,1,II}}$. The same values of frequencies $f_{\text{shear,1,II}}$ were also obtained
 7 including shear deformation in the FE model.

8 9 **4.2. Comparison with the FE model proposed by Jaiswal including shear deformation**

10 The frequencies ($f_{1,\text{exp}}$) were secondly compared with the FE model proposed by Jaiswal [9] in which, the
 11 post-tensioning force ($N_{0x,\text{aver}}$) is applied in the form of initial tension in the cables [17, 20]. Here $N_{0x,\text{aver}}$
 12 becomes an integral part of the thin-walled girder-cable system. As a result, $N_{0x,\text{aver}}$ is not treated as a
 13 compressive force, as takes place in Eq. (1) and Eq. (4). Elastic material, geometric properties and self-
 14 masses per unit length m_{beam} and m_{tendon} (Table 1 and Section 3.1) were accounted for the corresponding
 15 FE model in Strand7 [36]. Transverse shear deformation was considered, whilst longitudinal vibrations,
 16 warping and rotary inertia were neglected [37]. Particularly, the thin-walled box-girder was discretized
 17 into 20 elements, whilst the 5 strands were modeled using a straight tendon (A_{tendon}) in turn discretized
 18 into 2 truss elements [Fig. 7(a)]. The tendon was aligned to the girder axis and only had common nodes
 19 at the RHS ends. Here, the eccentricity (e) was the distance of the tendon from the girder's center-line, as
 20 shown in Fig. 7(a). In such FE model for steel members with an eccentric unbonded tendon, only end
 21 nodes of beam and tendon are connected by rigid links [Fig. 7(a)]. The mass of the hydraulic jack, $W =$
 22 118.5 N [Fig. 5(c)], was first accounted as a distributed load [$m_{\text{tot}} = m_{\text{beam}} + (W/L)$]. Second, the jack was
 23 accounted as a concentrated mass (W) modeled as a short beam element and connected at the midspan
 24 [Fig. 7(a)]. To obtain the FE frequencies ($f_{\text{E.-B.,1,II,FE}}$, $f_{\text{shear,1,II,FE}}$, $f_{\text{E.-B.,1,II,FE,W}}$, $f_{\text{shear,1,II,FE,W}}$), a two-step
 25 approach was performed. The post-tensioning force ($N_{0x,\text{aver}}$) was applied in the tendon, while axial
 26 deformations and second-order curvatures of the systems were gained. Artificial geometric imperfections
 27 were not included to prevent favoring instability modes. Then, the FE frequencies were evaluated for the
 28 corresponding new deformed configurations. Notably, the cross sectional area of strands (A_{tendon}) was

1 assumed as unchanged after deformations. Vice versa, the deformed configurations related to the first-
 2 order FE frequencies ($f_{E-B.,1,I,FE}$, $f_{shear,1,I,FE}$, $f_{E-B.,1,I,FE,W}$, $f_{shear,1,I,FE,W}$) were only caused by the self-masses of
 3 girder, tendon and jack (W). Yet, an additional FE model was generated in which the tendon (A_{tendon}) was
 4 connected to the surrounding RHS by one rigid link at the midspan, as depicted in Fig. 7(b), thus
 5 simulating the presence of a deviator. Figs. 7(a)–(b) show the two fundamental flexural shapes $v^{(0)}$ of the
 6 thin-walled box-girder with concentrated mass (W) at the midspan: it is clearly visible that only the RHS
 7 ends are governed by the conservation of the planarity of the cross sections.

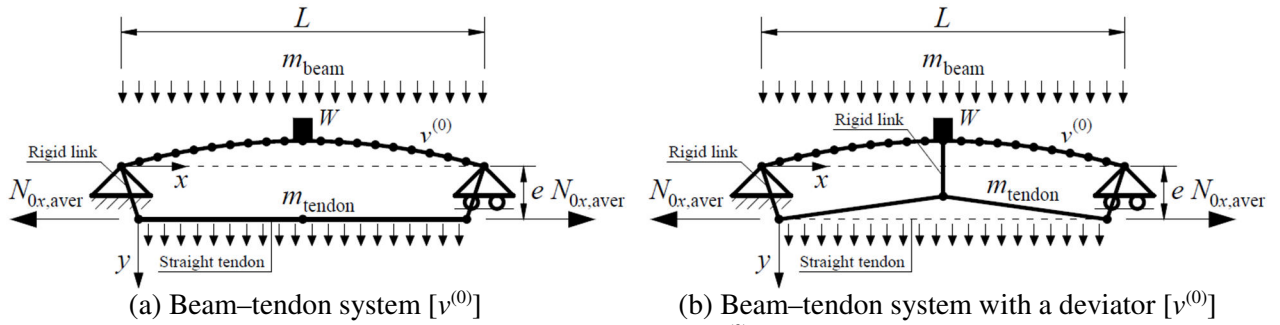


Fig. 7. FE model [9]. Fundamental flexural shapes $v^{(0)}$ according to the equivalent beam-tendon systems with mass W . The dashed lines represent the undeformed configurations.

8
 9 The comparisons are listed in Table 3; percentage errors $\Delta = (f_{1,FE} - f_{1,exp}) / f_{1,exp}$. When the total
 10 self-mass per unit length (m_{tot}) was accounted, the FE frequencies $f_{E-B.,1,I,FE}$ decreased from ≈ 18.95 to
 11 18.50 Hz into the increment $\approx 300 \leq N_{0x,aver} \leq 598$ kN, corresponding to an average error $\Delta_{aver} = 5.8\%$,
 12 whereas $f_{shear,1,I,FE}$ decreased from ≈ 18.84 to 18.38 Hz into the same range, corresponding to an error Δ_{aver}
 13 $= 5.2\%$. Conversely, when the jack was accounted as a mass (W), the FE frequencies $f_{E-B.,1,I,FE,W}$ decreased
 14 from ≈ 18.73 to 18.29 Hz into $\approx 300 \leq N_{0x,aver} \leq 598$ kN, corresponding to an error $\Delta_{aver} = 4.6\%$, whereas
 15 $f_{shear,1,I,FE,W}$ decreased from ≈ 18.62 to 18.17 Hz, corresponding to an error $\Delta_{aver} = 4.0\%$. The experimental
 16 frequencies ($f_{1,exp}$) followed the decreasing tendency of the aforementioned FE frequencies (Table 3).
 17 Consequently, they disagreed with findings and deductions of Noble et al. [14]. Similarly to what
 18 predicted by the reference solutions (Section 4.1), the slight initial increment in frequency ($f_{1,exp}$) of 0.02
 19 Hz was not of the same rate. Generally, the FE model, which included shear deformation, and accounted
 20 for the jack as a mass (W) at the midspan ($f_{shear,1,I,FE,W}$), better simulated the compression-softening effects.
 21 Table 3 also displays the trend of FE frequency ($f_{shear,1,I,FE,W,D}$) with the presence of a deviator [Fig. 7(b)].
 22 In this configuration, the compression-softening effect is eliminated when the tendon is in contact with

1 the surrounding thin-walled section: $f_{\text{shear},1,II,FE,W,D}$ slightly decreased from 18.52 to 18.34 Hz into the
2 increment $0 \leq N_{0x,aver} \leq 598$ kN. Notably, the FE model considered the slight influence of the bending
3 moments at the pinned-end supports (Fig. 4), $M_{\mu} = \mu_{\text{sliding}} R \times h / 2 = \mu_{\text{sliding}} \{ [(m_{\text{tot}} + m_{\text{tendon}}) \times L] / 2 \} \times h /$
4 2, caused by the dynamic sliding friction between steel-on-steel. A frictional coefficient (μ_{sliding}) of 0.42,
5 regarding the contact between dry steel-on-steel surfaces, was assumed [38].

6 The correlations summarized in Tables 2–3 confirmed the use of an accurate value of fundamental
7 frequency ($f_{1,\text{exp}}$) as indicator for axial load estimation in steel beams, bridge cables and hangers [34, 39–
8 42]. Yet, this work made certain such use for post-tensioning loss identification in thin-walled steel-
9 girder-bridges with straight cables that are not in contact with the surrounding section [1, 43, 44]. In fact,
10 second-order effects are not negligible when transverse vibrations are induced: a decrement in post-
11 tensioning force ($N_{0x,aver}$) of 100 kN corresponds to an increment in $f_{1,\text{exp}}$ of ≈ 0.18 Hz. By contrast, when
12 the cables are in contact with the section or draped with a deviator [43, 45, 46], its use as a parameter was
13 made doubtful: a decrement in $N_{0x,aver}$ of 100 kN corresponds to an increment in $f_{1,\text{exp}}$ of 0.03 Hz.
14 Regarding PC girder-bridges, the frequency ($f_{1,\text{exp}}$) was instead deduced to be a suitable indicator for
15 flexural rigidity determination because the dynamics of a pre or post-tensioning force is experimentally
16 negligible with a straight tendon [11, 47, 48] whilst is almost neglectable with a parabolic one [8, 12, 16].

17 18 **4.3. Comparison with FE plate and shell modeling**

19 FE plate and shell modeling was thirdly conducted to numerically compare the frequencies ($f_{1,\text{exp}}$). The
20 spatial solution of the thin-walled steel-box-girder was discretized into a group of FE linked to each other
21 in a certain manner. Since basic elements can be combined in different ways, FE analyses can solve
22 problems of dynamics of thin-walled structures involving complicated geometries and/or boundary
23 conditions [49, 50]. Elastic material, geometric properties, self-masses and unit weights (Table 1 and
24 Section 3.1) were accounted in Strand7 [36]. The thin-walled box-girder was modeled using 276,860
25 plate/shell elements with a mesh size=5×5 mm². The 5 straight strands were discretized into 2 truss
26 elements per each and fixed at the RHS ends according to the member design [Fig. 3(b)]. The plates for
27 transducers (24,000 elements), internal stiffeners (6,948 elements) and plates at the RHS ends (3,846
28 elements) were also modeled using plate/shell elements with the same mesh size. The upper part of

1 supports of 100 mm in width, 60 mm in height and 141 mm in length (13,536 elements), load cells (24,422
2 elements) and circular plates (12,211 elements) were instead modeled using brick elements with a mesh
3 size= $5 \times 5 \times 5$ mm³. The meshes between thin-walled girder, strands, plates and load cells were rigidly
4 connected at the RHS ends. The seismometers A0, A1, A2 and A3 and the jack (W) [Fig. 5(c)] were
5 modeled as short beams and connected along the girder according to the test layout (Fig. 4). Young's
6 modulus of supports, load cells, circular plates, seismometers and jack (W) was assumed of 200 GPa,
7 whilst $\nu = 0.25$. Carriage-hinge external constraints, to have the pinned-end conditions, were arranged
8 on a row of nodes at L (Fig. 4) whilst longitudinal vibrations, warping and rotary inertia were neglected.
9 To obtain the FE frequencies ($f_{1,II,FE,P\&S}$), the two-step procedure was conducted (Section 4.2). The post-
10 tensioning force ($N_{0x,aver}$) was applied in the strands and the axial deformation of the girder-tendon system
11 was gained. Artificial geometric imperfections were not included. Subsequently, $f_{1,II,FE,P\&S}$ were achieved.
12 Yet, an additional FE model was created to simulate the contact between strands and surrounding RHS
13 using one deviator, of 6 mm in thickness, at the midspan. 1,923 plate/shell elements with a mesh size= 5×5
14 mm² were adopted ($\rho_{beam} = 77.01$ kN/m³; $E = 205.12$ GPa) [Fig. 8(b)]. In this case, the FE frequencies
15 were labeled as $f_{1,II,FE,P\&S,D}$. Fig. 8(a) and Fig. 8(c) respectively show the FE fundamental flexural shape
16 of the thin-walled box-girder without and with deviator for $N_{0x,aver,max} = 598$ kN.

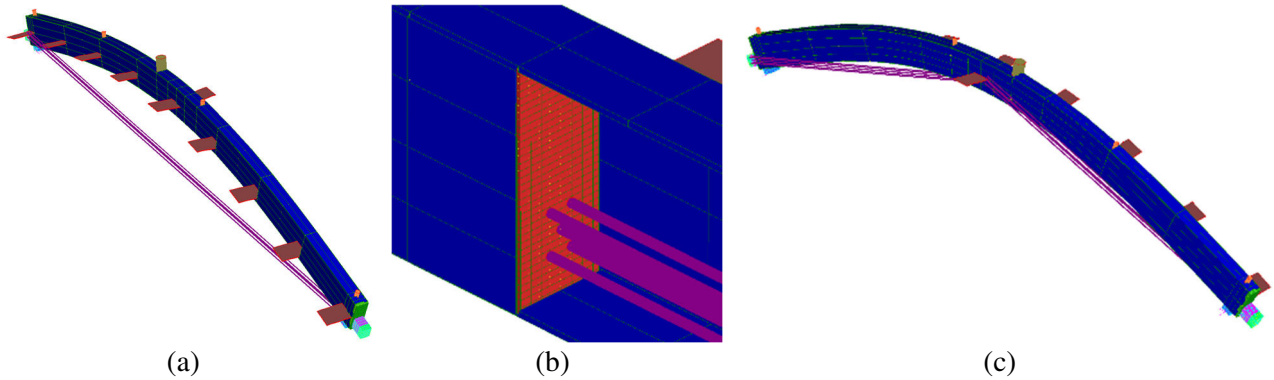


Fig. 8. FE plate/shell modeling in Strand7. Thin-walled box-girder (Fig. 4): (a) Fundamental flexural shape for $N_{0x,aver,max}=598$ kN; (b) Deviator at the midspan; (c) Fundamental flexural shape with deviator ($N_{0x,aver,max}=598$ kN).

17 Comparisons were reported in Table 2 [$\Delta = (f_{1,FE,P\&S} - f_{1,exp}) / f_{1,exp}$]. $f_{1,II,FE,P\&S}$ decreased from ≈ 18.61 to
18 18.14 Hz corresponding to an average error $\Delta_{aver} = 3.8\%$. Therefore, the dynamics of the thin-walled box-
19 girder was numerically well described since the experimental frequencies ($f_{1,exp}$) properly followed the
20 decreasing tendency of the compression-softening theory into the range $\approx 300 \leq N_{0x,aver} \leq 598$ kN.
21

1 Conversely, $f_{I,II,FE,P\&S,D}$ slightly decreased from 18.51 to 18.33 Hz into the increment $0 \leq N_{0x,aver} \leq 598$ kN
2 (Table 2). It was confirmed that the compression-softening effect is canceled when the cables are in
3 contact with the surrounding thin-walled section (Section 4.2). Likewise, the FE plate and shell modeling
4 always considered the dynamic sliding friction ($\mu_{sliding} R$) at the girder's pinned-end supports (Fig. 4)
5 because of the contact between dry steel-on-steel surfaces.

6 **Table 2.** Comparison between measured ($f_{1,exp}$) and numerical fundamental frequencies from
7 reference solutions [Eqs. (1), (2) and (4)] and FE plate and shell modeling. Seismometers **A1** and **A2** for BS
8 = 65,536 samples (Fig. 4).

N_{0x1} (kN)	N_{0x2} (kN)	$N_{0x,aver}$ (kN)	$f_{1,exp}$ (Hz)	Reference Solutions			FE Plate and Shell Modeling				
				Euler–Bernoulli		Shear		Shear		Shear with a Deviator	
				II–ord.	I–ord.	II–ord.	I–ord.	II–ord.	I–ord.	II–ord.	
–	–	0	–	$f_{E.-B.,I,II}$ Eq. (1) (Hz)	$f_{shear,I,I}$ Eq. (2) (Hz)	$f_{shear,I,II}$ Eq. (4) (Hz)	$f_{I,I,FE,P\&S}$ (Hz)	$f_{I,II,FE,P\&S}$ (Hz)	$f_{I,I,FE,P\&S,D}$ (Hz)	$f_{I,II,FE,P\&S,D}$ (Hz)	
–	–	0	–	19.50	19.39	–	19.18	–	18.51	–	
111	122	117	18.37	19.30 (5.1%)	19.39 (5.6%)	19.19 (4.5%)	19.18 (4.4%)	18.98 (3.3%)	18.51	18.47	
205	222	214	18.39	19.12 (4.0%)	19.39 (5.4%)	19.01 (3.4%)	19.18 (4.3%)	18.82 (2.3%)	18.51	18.45	
323	341	332	17.94	18.91 (5.4%)	19.39 (8.1%)	18.80 (4.8%)	19.18 (6.9%)	18.61 (3.7%)	18.51	18.41	
410	432	421	17.78	18.75 (5.5%)	19.39 (9.1%)	18.64 (4.8%)	19.18 (7.9%)	18.45 (3.8%)	18.51	18.38	
490	512	501	17.64	18.60 (5.4%)	19.39 (9.9%)	18.49 (4.8%)	19.18 (8.7%)	18.31 (3.8%)	18.51	18.36	
586	610	598	17.46	18.42 (5.5%)	19.39 (11.1%)	18.32 (4.9%)	19.18 (9.9%)	18.14 (3.9%)	18.51	18.33	
$m_{tot} = m_{beam} + (W/L)$							m_{beam} and $W \rightarrow$ Mass W concentrated				

9
10 **Table 3.** Comparison between measured ($f_{1,exp}$) and numerical fundamental frequencies from the FE
11 model [9]. Seismometers **A1** and **A2** for BS = 65,536 samples (Fig. 4).

$N_{0x,aver}$ (kN)	$f_{1,exp}$ (Hz)	FE Model proposed by Jaiswal									
		Euler–Bernoulli		Shear		Euler–Bernoulli		Shear		Shear with a Deviator	
		I–ord.	II–ord.	I–ord.	II–ord.	I–ord.	II–ord.	I–ord.	II–ord.	I–ord.	II–ord.
0	–	$f_{E.-B.,I,FE}$ (Hz)	$f_{E.-B.,II,FE}$ (Hz)	$f_{shear,I,FE}$ (Hz)	$f_{shear,II,FE}$ (Hz)	$f_{E.-B.,I,FE,W}$ (Hz)	$f_{E.-B.,II,FE,W}$ (Hz)	$f_{shear,I,FE,W}$ (Hz)	$f_{shear,II,FE,W}$ (Hz)	$f_{shear,I,FE,W,D}$ (Hz)	$f_{shear,II,FE,W,D}$ (Hz)
0	–	19.50	–	19.39	–	19.28	–	19.17	–	18.52	–
117	18.37	19.50 (6.2%)	19.31 (5.1%)	19.39 (5.6%)	19.20 (4.5%)	19.28 (5.0%)	19.09 (3.9%)	19.17 (4.4%)	18.98 (3.3%)	18.52	18.49
214	18.39	19.50 (6.0%)	19.15 (4.1%)	19.39 (5.4%)	19.04 (3.5%)	19.28 (4.8%)	18.93 (2.9%)	19.17 (4.2%)	18.82 (2.3%)	18.52	18.46
332	17.94	19.50 (8.7%)	18.95 (5.6%)	19.39 (8.1%)	18.84 (5.0%)	19.28 (7.5%)	18.73 (4.4%)	19.17 (6.9%)	18.62 (3.8%)	18.52	18.42
421	17.78	19.50 (9.7%)	18.80 (5.7%)	19.39 (9.1%)	18.69 (5.1%)	19.28 (8.4%)	18.59 (4.6%)	19.17 (7.8%)	18.47 (3.9%)	18.52	18.40
501	17.64	19.50 (10.5%)	18.66 (5.8%)	19.39 (9.9%)	18.55 (5.2%)	19.28 (9.3%)	18.45 (4.6%)	19.17 (8.7%)	18.34 (4.0%)	18.52	18.37
598	17.46	19.50 (11.7%)	18.50 (6.0%)	19.39 (11.1%)	18.38 (5.3%)	19.28 (10.4%)	18.29 (4.8%)	19.17 (9.8%)	18.17 (4.1%)	18.52	18.34
$m_{tot} = m_{beam} + (W/L)$							m_{beam} and $W \rightarrow$ Mass W concentrated				

1 **5. Experiments performed by Noble et al.**

2 **5.1. Free transverse vibration of two thin-walled steel-box members post-tensioned by a concentric** 3 **straight cable**

4 In Noble et al. [14], the assumption that a post-tensioning force is whether or not dynamically equivalent
5 to a compressive force was investigated (Section 1). To make sure that findings and deductions on our
6 tests were effective (Section 4), numerical validations of the experiments performed by Noble et al. [14]
7 were additionally executed. Specifically, the researchers carried out free transverse vibrations on two thin-
8 walled steel members with RHS post-tensioned by a concentric straight cable. Beam 1 had $b = 50$ mm in
9 width, $h = 30$ mm in height, and $t = t_1 = 3$ mm in thickness, whereas Beam 2 had $b = 120$ mm, $h = 60$ mm,
10 and $t = t_1 = 3$ mm. Their total self-masses per unit length ($m_{\text{tot}} = \rho_{\text{beam}} \times A$) were 0.0343 and 0.0806 kN/m,
11 respectively. The steel-box members were post-tensioned by a concentric steel “seven-wire” strand of
12 15.7 mm in diameter, whereas their material and geometric properties were summarized in Table 4. The
13 self-mass per unit length of the strand ($m_{\text{tendon}} = \rho_{\text{tendon}} \times A_{\text{tendon}}$) was instead 0.0114 kN/m. It was anchored
14 with collets either side of two hydraulic jacks of 30 tons. in force capacity, and 222.7 N in mass per each
15 (RCD307, Simplex). A load cell of 85 mm in diameter, 80 mm in length, 300 kN in force capacity, 1.5
16 mV/V in sensitivity, and 19.6 N in mass, was instead fixed at the extremities of the two jacks to measure
17 the applied post-tensioning forces ($N_{0x,\text{aver}}$) (KCM-300KNA, Tokyo Measuring Instruments Lab.).
18 Moreover, a $200 \times 150 \times 20$ mm³ steel plate, of 77.01 kN/m³ in unit weight, 200 GPa in Young’s modulus,
19 and 80 GPa in shear one, was positioned between each jack and the RHS end. Also, the members were
20 pinned-end by steel knife-edge supports in a test rig, resulting in a clear length L (Table 4). One of the
21 jacks was connected to a hydraulic hand pump to apply a post-tensioning force ($N_{0x,\text{aver}}$) by pulling the
22 strand outward. Thus, impact hammer testing were executed at incremental levels of $N_{0x,\text{aver}}$ [Fig. 3 in
23 Noble et al. [14]]. In Beam 1, the maximum induced second-order effects were $\approx 89\%$ of its critical
24 buckling load $N_{\text{crE},1}$ or $N_{\text{crE, shear},1} = 56$ kN. Differently, in Beam 2, maximum second-order effects were \approx
25 30% of its $N_{\text{crE},1}$ or $N_{\text{crE, shear},1} = 597$ kN. The strand was only in contact at the edges, but not throughout
26 the RHS length during vibrations. Therefore, the eccentricity (e) of $N_{0x,\text{aver}}$ was theoretically zero (Fig. 1).
27 Following execution of hammer testing, acceleration-time data, with respect to the specimens’ horizontal
28 axis, were acquired by an accelerometer of 6 g in mass, and $0.05 \text{ mV}(\text{m/s}^2)^{-1}$ in sensitivity, which was

1 vertically arranged at the midspans (3200B4, Dytran Instruments) [Figs. 7 and 9 in Noble et al. [14]]. The
2 FFT was then performed on the acceleration-time data to describe the signal in the frequency domain.
3 And, peak picking method was used to identify the fundamental frequencies ($f_{1,exp}$) (Section 3.4). The
4 frequencies ($f_{1,exp}$) were determined for each increment of post-tensioning force ($N_{0x,aver}$) and each hammer
5 test conducted [Figs. A1(b) and A1(d) in Noble et al. [14]]. More information on experiments and modal
6 analyses are reported in Noble et al. [14]. The applied post-tensioning forces ($N_{0x,aver}$) and average
7 identified frequencies ($f_{1,exp}$) were also listed in Table 5. For both specimens, a slight decreasing trend in
8 frequency ($f_{1,exp}$) was observed, which was not of the same rate as predicted by the Euler-Bernoulli theory
9 [Eq. (1)]. In Beam 1, $f_{1,exp}$ decreased from 27.77 to 24.77 Hz into the increment in post-tensioning force
10 ($N_{0x,aver}$) which went from 0 to 50 kN. Vice versa, in Beam 2, $f_{1,exp}$ decreased from 75.35 to 72.85 Hz into
11 an increment which went up to 180 kN. Regression lines were fitted to the data ($f_{1,exp}$) and depicted in
12 blue in Figs. 9(a)–(b). Accordingly, Noble et al. [14] concluded that a post-tensioning force is not
13 dynamically equivalent to a compressive force. Strictly, the researchers deduced that the compression-
14 softening theory is not applicable to prestressed structures. Based on our findings and deductions (Section
15 4), it is questionable that the frequencies ($f_{1,exp}$) of their post-tensioned box members did not significantly
16 follow the decreasing trend of the compression-softening theory [Figs. A1(b) and A1(d) in Noble et al.
17 [14]].

18
19

Table 4. Noble et al. [14]. Properties of Beam 1 and Beam 2.

Property	Beam 1) RHS $50 \times 30 \times 3 \text{ mm}^3$	Beam 2) RHS $120 \times 60 \times 3 \text{ mm}^3$
$I \text{ (mm}^4\text{)}$	6.1812×10^4	6.64092×10^5
$A \text{ (mm}^2\text{)}$	444	1,044
$A_w \text{ (mm}^2\text{)}$	144	324
Slenderness	127	59
k [Eq. (3)]	0.321	0.251
$L \text{ (m)}$		1.5
$\rho_{beam} \text{ (kN/m}^3\text{)}$		77.20
$E \text{ (GPa)}$		205
$f_{yk} \text{ (MPa)}$		235
$G \text{ (GPa)}$		82
ν		0.25
$A_{tendon} \text{ (mm}^2\text{)}$		149
$\rho_{tendon} \text{ (kN/m}^3\text{)}$		76.62
$E_{tendon} \text{ (GPa)}$		200
$\sigma_{ty} \text{ (MPa)}$		1,860

20
21

1 **5.2. Comparison with reference solutions**

2 Following the numerical approaches described in Section 4, Noble et al. [14]’s fundamental frequencies
 3 ($f_{1,exp}$) were firstly compared with Eq. (1) and Eq. (4) in which $n = 1$, whereas the critical buckling loads
 4 of Beams 1 and 2, $N_{crE,1}$ and $N_{crE, shear,1}$, were assumed (Section 2). Comparisons were presented in Figs.
 5 9(a)–(b): the trends of frequencies $f_{E-B,1,II}$ were depicted in pink [Eq. (1)], whereas those of frequencies
 6 $f_{shear,1,II}$ were depicted in red with cross symbols [Eq. (4)]. In Beam 1, $f_{E-B,1,II}$ decreased from 42.04 to
 7 13.32 Hz into the increment in post-tensioning force ($0 \leq N_{0x,aver} \leq 50$ kN), whereas $f_{shear,1,II}$ decreased
 8 from 41.94 to 13.29 Hz [Fig. 9(a)]. In Beam 2, $f_{E-B,1,II}$ decreased from 89.87 to 75.11 Hz into the increment
 9 $0 \leq N_{0x,aver} \leq 180$ kN, whereas $f_{shear,1,II}$ decreased from 88.64 to 74.09 Hz [Fig. 9(b)]. Therefore, the
 10 frequencies $f_{1,exp}$, underlined by the blue regression lines, did not follow the significant decreasing
 11 tendency of the compression-softening theory of Eq. (1) and Eq. (4). The experimental behavior of the
 12 post-tensioned steel-box members was not well described confirming findings and deductions of Noble
 13 et al. [14]. The considerable softening effect, in terms of second-order effects, is particularly visible in
 14 Beam 1 ($\approx 45\%$ of its $N_{crE,1}$ or $N_{crE, shear,1}$). Indeed, the frequencies $f_{E-B,1,II}$ and $f_{shear,1,II}$ started to converge
 15 to the buckling loads ($N_{crE,1}$ and $N_{crE, shear,1}$) at a level of post-tensioning force ($N_{0x,aver}$) of ≈ 25 kN. Notably
 16 also that, in Fig. 9(a), shear deformation is visibly negligible due to the high slenderness of Beam 1 equal
 17 to 127.

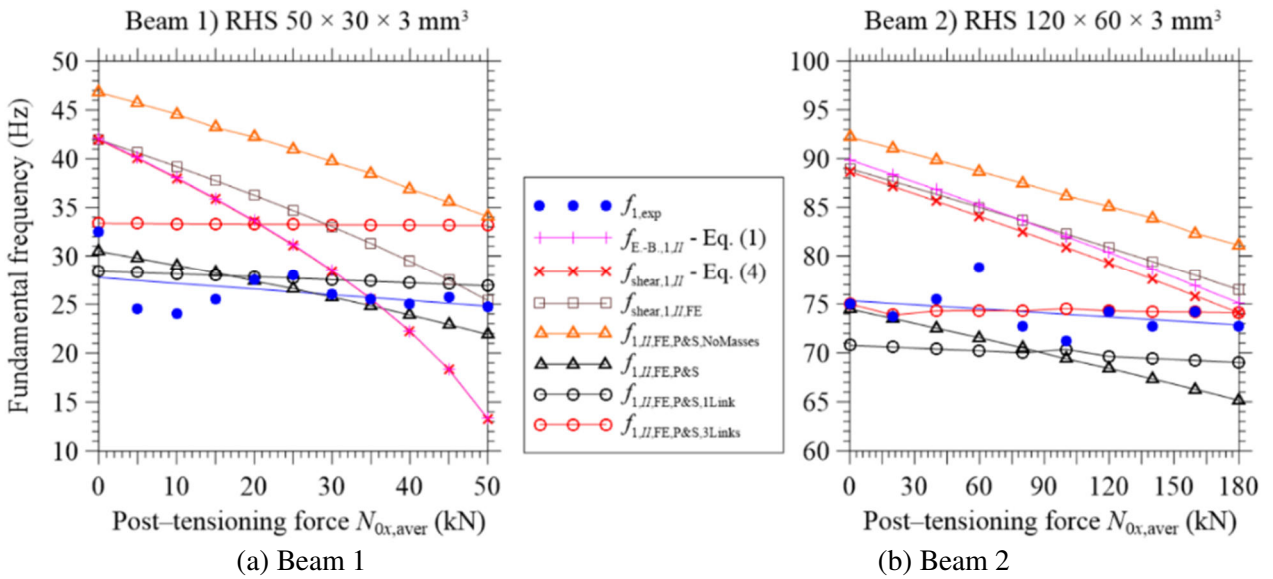


Fig. 9. Noble et al. [14]. Comparison between experimental ($f_{1,exp}$) and numerical fundamental frequencies from reference solutions [Eqs. (1) and (4)] and FE models.

1
2

Table 5. Noble et al. [14]. Comparison between experimental ($f_{1,exp}$) and numerical fundamental frequencies from FE plate and shell modeling.

Beam 1) RHS $50 \times 30 \times 3 \text{ mm}^3$					Beam 2) RHS $120 \times 60 \times 3 \text{ mm}^3$				
		Shear	Shear with 1 Rigid Link	Shear with 3 Rigid Links			Shear	Shear with 1 Rigid Link	Shear with 3 Rigid Links
		II-ord.	II-ord.	II-ord.			II-ord.	II-ord.	II-ord.
$N_{0x,aver}$ (kN)	$f_{1,exp}$ (Hz)	$f_{1,II,FE,P\&S}$ (Hz)	$f_{1,II,FE,P\&S,1Link}$ (Hz)	$f_{1,II,FE,P\&S,3Links}$ (Hz)	$N_{0x,aver}$ (kN)	$f_{1,exp}$ (Hz)	$f_{1,II,FE,P\&S}$ (Hz)	$f_{1,II,FE,P\&S,1Link}$ (Hz)	$f_{1,II,FE,P\&S,3Links}$ (Hz)
0	32.50 –	30.49 (-6.2%)	28.42 (-12.6%)	33.37 (2.7%)	0	75.00 –	74.48 (-0.7%)	70.78 (-5.6%)	75.00 (0.0%)
5	24.50 –	29.76 (21.5%)	28.27 (15.4%)	33.35 (36.1%)	20	73.75 –	73.52 (-0.3%)	70.59 (-4.3%)	73.87 (0.2%)
10	24.00 –	29.02 (20.9%)	28.13 (17.2%)	33.32 (38.8%)	40	75.50 –	72.52 (-3.9%)	70.39 (-6.8%)	74.29 (-1.6%)
15	25.50 –	28.26 (10.8%)	27.98 (9.7%)	33.30 (30.6%)	60	78.75 –	71.56 (-9.1%)	70.22 (-10.8%)	74.33 (-5.6%)
20	27.50 –	27.39 (-0.4%)	27.83 (1.2%)	33.28 (21.0%)	80	72.75 –	70.52 (-3.1%)	70.00 (-3.8%)	74.30 (2.1%)
25	28.00 –	26.59 (-5.0%)	27.68 (-1.1%)	33.26 (18.8%)	100	71.25 –	69.39 (-2.6%)	70.36 (-1.2%)	74.53 (4.6%)
30	26.00 –	25.74 (-1.0%)	27.53 (5.9%)	33.23 (27.8%)	120	74.25 –	68.42 (-7.9%)	69.64 (-6.2%)	74.30 (0.1%)
35	25.50 –	24.84 (-2.6%)	27.37 (7.3%)	33.21 (30.2%)	140	72.75 –	67.35 (-7.4%)	69.43 (-4.6%)	74.22 (2.0%)
40	25.00 –	23.90 (-4.4%)	27.22 (8.9%)	33.19 (32.8%)	160	74.25 –	66.27 (-10.7%)	69.22 (-6.8%)	74.17 (-0.1%)
45	25.75 –	22.92 (-11.0%)	27.06 (5.1%)	33.16 (28.8%)	180	72.75 –	65.16 (-10.4%)	69.02 (-5.1%)	74.10 (1.9%)
50	24.75 –	21.89 (-11.6%)	26.90 (8.7%)	33.14 (33.9%)					

3

4 **5.3. Comparison with the FE model proposed by Jaiswal including shear deformation**

5 Noble et al. [14]’s fundamental frequencies ($f_{1,exp}$) were secondly compared with the FE model proposed
6 by Jaiswal [9] which was created following the approaches described in Section 4.2. Elastic material,
7 geometric properties and self-masses per unit length (m_{tot} and m_{tendon}) of the two post-tensioned box
8 members (Table 4 and Section 5.1) were accounted for the corresponding FE models in Strand7 [36].
9 Particularly, the straight cable was aligned to the center-line of the members, and only connected to the
10 end nodes of the RHS as shown in Fig. 10(a). Besides, Beams 1 and 2 were discretized into 20 elements,
11 whilst cables were discretized into 2 truss elements [Fig. 10(a)]. To obtain the FE frequencies ($f_{shear,1,II,FE}$),
12 the two-step approach was performed. The post-tensioning force ($N_{0x,aver}$) was applied in the cable and
13 the axial static deformation of each system was achieved. Then, the FE frequencies ($f_{shear,1,II,FE}$) were
14 gained for the corresponding new deformed configurations. Notably, the cross sectional area of cables
15 (A_{tendon}) was assumed as unchanged after deformations. Fig. 10(b) shows the FE fundamental flexural

1 shape of Beam 2 for $N_{0x,aver,max}=180$ kN [Fig. 9(b)]: it is clearly visible that the cable remains horizontal,
 2 thus only the ends of the member–cable system are governed by the conservation of the planarity of the
 3 cross sections.

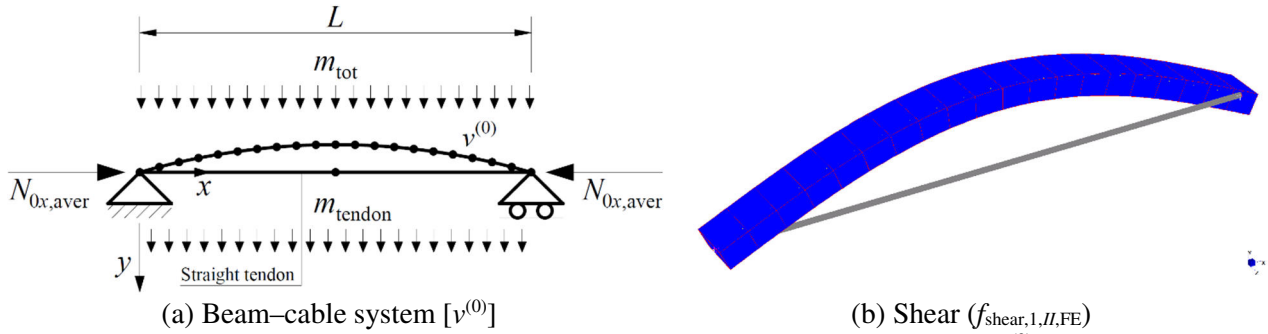


Fig. 10. Noble et al. [14]. FE model [9]. Fundamental flexural shapes: (a) $v^{(0)}$ according to the equivalent beam–cable system; (b) Beam 2 for $N_{0x,aver,max}=180$ kN in Strand7.

4 The comparisons were presented in Figs. 9(a)–(b): the trends of FE frequencies $f_{\text{shear},1,II,FE}$ were
 5 depicted in brown. In Beam 1, $f_{\text{shear},1,II,FE}$ decreased from 41.96 to 25.39 Hz into the increment $0 \leq N_{0x,aver}$
 6 ≤ 50 kN [Fig. 9(a)]. Conversely, in Beam 2, $f_{\text{shear},1,II,FE}$ decreased from 88.96 to 76.47 Hz into $0 \leq N_{0x,aver} \leq$
 7 180 kN [Fig. 9(b)]. We see again that the frequencies $f_{1,exp}$ did not follow the decreasing tendency of
 8 $f_{\text{shear},1,II,FE}$. Accordingly, the dynamics of the two post–tensioned members was not properly described.
 9 The softening effects were lower than those simulated by Eq. (1) and Eq. (4) because the cables increased
 10 the stiffness of each member–cable system. This behavior was extremely evident in Beam 1 having a high
 11 slenderness [Fig. 9(a)]. Notably, the FE models always considered the sliding friction ($\mu_{\text{sliding}} R$) at the
 12 members' supports because of the contact between dry steel-on-steel surfaces (Section 4.2).

14 5.4. Comparison with FE plate and shell modeling

15 FE plate and shell modeling was thirdly conducted to numerically compare Noble et al. [14] 's frequencies
 16 ($f_{1,exp}$). The FE approaches illustrated in Section 4.3 were adopted in Strand7 [36]. Each cable was
 17 discretized into 2 truss elements. Beams 1 and 2 were respectively modeled using 7,680 and 17,280
 18 plate/shell elements corresponding to a mesh size of 5×6.25 mm². Instead, the two steel plates were
 19 modeled using 9,600 brick elements for a mesh size= $5 \times 5 \times 5$ mm³. The meshes of thin-walled box
 20 members, cables and plates were rigidly connected at the RHS ends. Unit weights and elastic material
 21 properties were described in Table 4 and Section 5.1. Carriage-hinge external constraints were arranged
 22 on a row of nodes at L (Table 4). To obtain the FE frequencies ($f_{1,II,FE,P\&S,NoMasses}$), the two-step procedure
 23

1 was similarly conducted. The post-tensioning force ($N_{0x,aver}$) was applied in the cable and the axial
 2 deformation of each system was gained. Subsequently, the FE frequencies ($f_{1,II,FE,P\&S,NoMasses}$) were
 3 achieved. In Beam 1, $f_{1,II,FE,P\&S,NoMasses}$ decreased from 46.82 to 34.07 Hz into the increment $0 \leq N_{0x,aver} \leq$
 4 50 kN [in orange in Fig. 9(a)]. Vice versa, in Beam 2, $f_{1,II,FE,P\&S,NoMasses}$ decreased from 92.20 to 81.11 Hz
 5 into $0 \leq N_{0x,aver} \leq 180$ kN [in orange in Fig. 9(b)]. The softening effects were affected by the steel plates at
 6 the RHS ends which increased the stiffness of each member-cable system.

7 The influence of load cells and hydraulic jacks was afterwards investigated: Bonopera et al. [11,
 8 12] preliminarily declared that Noble et al. [14] 's frequencies ($f_{1,exp}$) were affected by such masses
 9 (Section 1). The load cells (KCM-300KNA), of 60 mm in width, 75 mm in height, 80 mm in length,
 10 modeled using 5,760 brick elements, for a mesh size= $5 \times 5 \times 5$ mm³, were accounted with their proper mass
 11 (Section 5.1). Conversely, the hydraulic jacks (RCD307), of 100 mm in width, 90 mm in height, 220 mm
 12 in length, modeled using 31,680 brick elements, and with the same mesh size, were accounted with a
 13 lower mass. In fact, it was found that the jacks were arranged touching the floor at their mid length [Fig.
 14 6.16. in Noble [51]]. Accordingly, their mass was hypothetically reduced of 50% to decrease their
 15 dynamic contribution. Young's modulus of load cells and jacks was assumed of 200 GPa, whilst $\nu = 0.25$.
 16 The meshes between cables, plates, load cells and jacks were rigidly connected. The FE frequencies
 17 ($f_{1,II,FE,P\&S}$) were obtained using the two-step procedure. In Beam 1, $f_{1,II,FE,P\&S}$ decreased from 30.49 to
 18 21.89 Hz into $0 \leq N_{0x,aver} \leq 50$ kN [in black with triangles in Fig. 9(a)]. Vice versa, in Beam 2, $f_{1,II,FE,P\&S}$
 19 decreased from 74.48 to 65.16 Hz into $0 \leq N_{0x,aver} \leq 180$ kN [in black with triangles in Fig. 9(b)].
 20 Subsequently, the contact between straight cable and surrounding RHS of the two box members was
 21 additionally simulated. First, it was simplified by one rigid link, which was used to connect cable and
 22 RHS upper part at their midspans. Second, it was simplified by three rigid links equidistant fastened along
 23 their length (L). In the first condition, for Beam 1, average percentage error $\Delta_{aver,1Link} = (f_{1,II,FE,P\&S,1Link} -$
 24 $f_{1,exp}) / f_{1,exp} = 6.0\%$ [Fig. 9(a)], whilst, for Beam 2, $\Delta_{aver,1Link} = -5.5\%$ [Fig. 9(b)]. In the second condition,
 25 for Beam 1, average error $\Delta_{aver,3Links} = (f_{1,II,FE,P\&S,3Links} - f_{1,exp}) / f_{1,exp} = 27.4\%$ [Fig. 9(a)], whilst, for Beam
 26 2, $\Delta_{aver,3Links} = 0.4\%$ [Fig. 9(b)]. The errors (Δ) between experiments and FE plate and shell modeling for
 27 each $N_{0x,aver}$ were listed in Table 5. Figs. 11(a)–(f) show the FE fundamental flexural shapes of Beams 1

1 and 2 according to their maximum post-tensioning ($N_{0x,aver,max}$). Based on the above correlations, it was
 2 confirmed that the mass at one end constraint, when is greater than the self-mass of the girder themselves,
 3 significantly affects the vibrational dynamics. Moreover, in Noble et al. [14], the cables probably touched
 4 the surrounding RHS causing stiffening effects which, in turn, falsified the decreasing tendencies of the
 5 compression-softening theory [Figs. A1(b) and A1(d)]. Similarly to what gathered by the FE analyses of
 6 our experiments (Sections 4.2 and 4.3), we deduced that the dynamics of a compressive force and that of
 7 a post-tensioning is different only when the cables are in contact with the surrounding beam's section.
 8 Likewise, the FE modeling always considered the sliding friction ($\mu_{sliding} R$) at the members' supports
 9 because of the contact between dry steel-on-steel surfaces (Section 4.2).

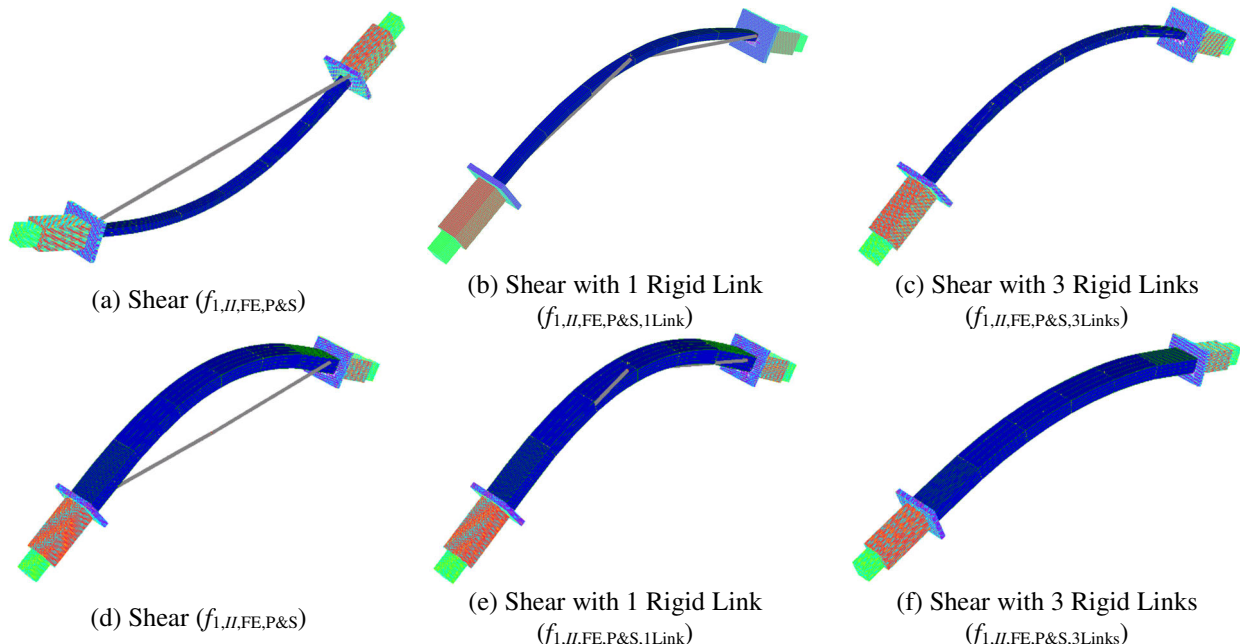


Fig. 11. Noble et al. [14]. FE plate/shell modeling in Strand7. Fundamental flexural shapes: (a–c) Beam 1 for $N_{0x,aver,max}=50$ kN; (d–f) Beam 2 for $N_{0x,aver,max}=180$ kN.

10

11

6. Relationship between free transverse vibration and critical buckling

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The effect of the location of straight cables on the relationship between free transverse vibrations and

13

critical buckling including shear deformation was discussed. In order to observe the evolution of

14

fundamental frequency in more detail, Fig. 12 and Fig. 13 compare the stability of thin-walled member-

15

cable systems [Fig. 7(a)] with that of the same girders when externally axially loaded (Fig. 1). In the

16

second case, the stability is determined by the critical force ($N_{0x,aver}$) for which the fundamental frequency

17

drops to zero [23]. For our thin-walled steel-box-girder (Fig. 2), the stability was numerically studied

18

since it was not possible to over post-tensioning the beam in the experimental set-up (Fig. 4). Fig. 12

1 demonstrates that increasing the post-tensioning force ($N_{0x,aver}$), the FE frequency $f_{shear,1,II,FE,W}$
2 considerably reduces up to $N_{0x,aver}$ of $\approx 1,000$ kN, corresponding to the eccentric post-tensioning force
3 ($N_{0x,ULS,0.23}$) which causes the yielding stress of the most extreme fibres of midspan cross section.
4 Subsequently, $f_{shear,1,II,FE,W}$ increases because of the stiffening effect due to the cables' eccentricity ($e/h =$
5 0.23), i.e., when the thin-walled girder is subjected to higher bending moments at the RHS ends ($N_{0x,aver}$
6 e) and second-order curvatures [21, 22]. Nonetheless, the eccentric post-tensioning force ($N_{0x,p1,0.23}$),
7 which causes the plasticity of the cross section, is equal to 1,484 kN. Conversely, when the eccentricity
8 of cables is large ($e/h = 0.74$), the stiffening effect generally prevails and the vibration tendon mode is
9 dominant. In this case, $N_{0x,ULS,0.74}$ and $N_{0x,p1,0.74}$ become respectively equal to 502 and 758 kN (Fig. 12).
10 Yet, when the corresponding initial second-order curvatures of the girder are not imposed as deformed
11 configurations (Section 4.2), $f_{shear,1,II,FE,W}$ follows the decreasing trend of the compression-softening theory
12 which, in turn, does not depend on cables' eccentricity (e), as occurred using FE plate and shell modeling
13 ($f_{1,II,FE,P\&S}$), Eq. (1) or Eq. (4). Specifically, the course of $f_{shear,1,II,FE,W}$ converges to a critical buckling load
14 of 5,890 kN, which is higher than $N_{crE,1}$ or $N_{crE,shear,1} = 5,543$ kN (Fig. 12), because of a stiffening effect
15 due to the formation of rotational elastic conditions in the cables at the RHS ends. Notably, post-tensioned
16 thin-walled steel-girder-bridges draped with a deviator [43, 45, 46] almost never experience critical
17 buckling phenomena [Fig. 7(b)] as commonly happened in PC girder-bridges [17].

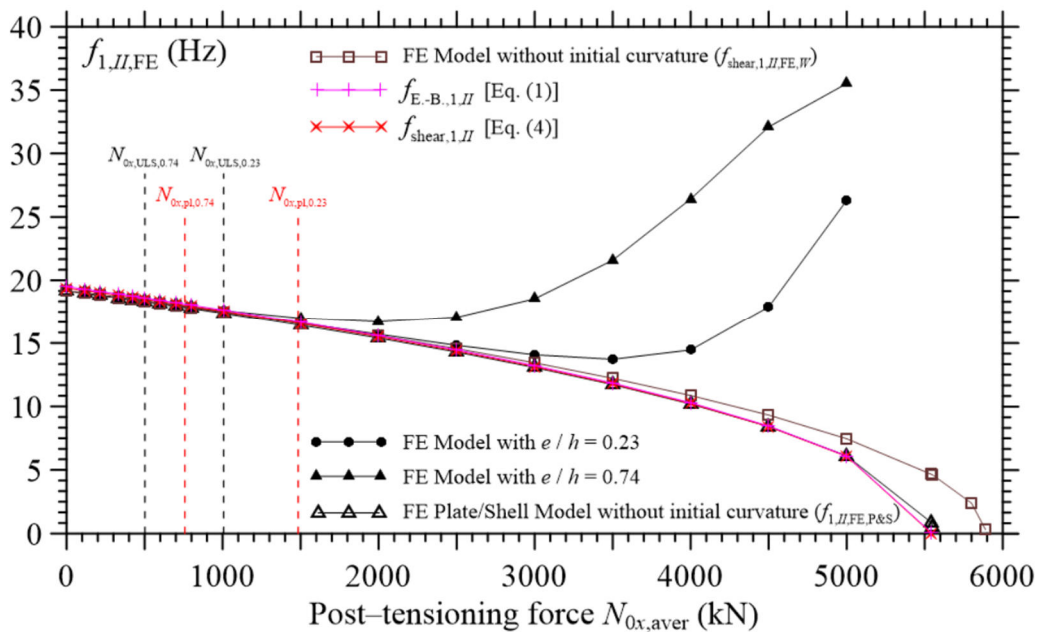


Fig. 12. Thin-walled member (Fig. 4). Effect of the post-tensioning force ($N_{0x,aver}$) on the fundamental frequency.

1 Fig. 13 shows the trend of $f_{\text{shear},1,II,FE}$ of the thin-walled steel-girder-bridge with unbonded cables
 2 utilized in the numerical study presented by Bai-jian et al. [1]. Likewise, increasing the post-tensioning
 3 force (N_{0x}) up to the critical buckling of the girder-cable system of 3,845 kN, $f_{\text{shear},1,II,FE}$ reduces the
 4 negative slope of its softening branch according to the cables' eccentricity (e), i.e., from small ($e/h=0.12$)
 5 to medium ($e/h=0.28$). The eccentric $N_{0x,ULS,0.12}$ and $N_{0x,pl,0.12}$ are respectively equal to 433 and 669 kN
 6 when $e/h=0.12$. Vice versa, $N_{0x,ULS,0.28}$ and $N_{0x,pl,0.28}$ become 314 and 483 kN when $e/h=0.28$ (Fig. 13).
 7 Thin-walled member-cable systems with an eccentrically straight tendon have two sets of fundamental
 8 vibrational modes, i.e., beam-dominated and tendon-dominated modes. Particularly, their dynamic
 9 interaction depends not only on the magnitude of post-tensioning force (N_{0x}) but also on the location of
 10 the tendon that induces an initial second-order curvature along the member. We can state that the vibration
 11 beam mode follows the compression-softening theory when tendon's eccentricity (e) is zero, and it is
 12 almost always dominant when the yielding stress under post-tensioning (N_{0x}) is not reached.

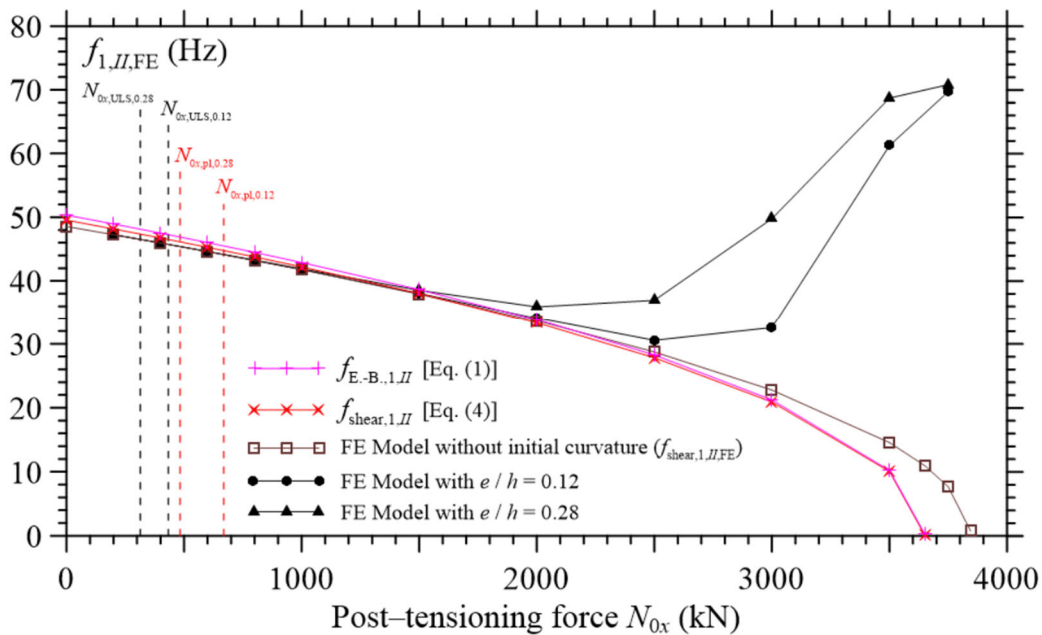


Fig. 13. Thin-walled girder-bridge [1]. Effect of the post-tensioning force (N_{0x}) on the fundamental frequency.

13
 14 **7. Conclusions**

15 The effect of pre and post-tensioning forces on the beam dynamics is object of a significant discrepancy.
 16 To solve this conflict, firstly, free transverse vibrations were induced to a post-tensioned thin-walled
 17 steel-box-girder in larger scale and with eccentric straight cables. Secondly, to finally investigate the
 18 validity of the compression-softening theory in prestressed beams, the experiments performed by Noble

1 et al. [14] were analyzed. Within the limitations of this research and the results thoroughly validated using
2 the bench-top experiment (Fig. 4), one proposed formula based on Timoshenko beam model [Eq. (4)],
3 and two high-fidelity FE models including shear deformation, the following conclusions are reported:

- 4 1. The experimental dynamics were appropriately simulated using the FE model proposed by Jaiswal
5 [9] and that based on plate and shell elements. In thin-walled member–cable systems, both
6 modeling apply the effective post–tensioning force ($N_{0x,aver}$) in the form of initial tension in the
7 cables whilst neglect the corresponding second-order curvature. Accordingly, the fundamental
8 frequency ($f_{1,exp}$) of thin-walled steel members is not influenced by the beam’s camber (initial
9 curvature) until the yielding stress (f_{yk}) under post–tensioning ($N_{0x,aver}$) is reached.
- 10 2. Conversely to what concluded by Noble et al. [14, 15], the dynamic effect of a compressive force
11 and that of a post–tensioning phenomenologically differ only when the cables are in contact with
12 the surrounding beam’s section [Fig. 7(b)]. Depending on this characteristic, the dynamics of all
13 forms of pre and post–tensioned members is described by compression-softening effect.
- 14 3. Similarly to what deduced for PC girder-bridges [7, 19, 20], the fundamental frequency ($f_{1,exp}$) as
15 parameter for post–tensioning loss identification is made doubtful in thin-walled steel-girder-
16 bridges when the cables touch the surrounding section or are draped with a deviator [Fig. 8(b)].
- 17 4. When the beam’s cambers (initial curvatures) are considered as deformed configurations within
18 the FE modeling, thin-walled member–cable systems have two sets of fundamental vibrational
19 modes (Figs. 12–13). By increasing the post–tensioning force ($N_{0x,aver}$) beyond the value causing
20 the yielding stress of the most extreme fibres of cross section (f_{yk}), a stiffening effect occurs. This
21 consequence depends on the straight tendon’s eccentricity (e): when it is small ($e/h = 0.12 \sim 0.23$)
22 the softening effect is prevalent than the stiffening one and the vibration beam mode is dominant.
23 Vice versa, when the eccentricity is large ($e/h = 0.44 \sim 0.74$) the stiffening effect prevails and the
24 vibration tendon mode is dominant (Figs. 12–13).

25

26 **CRedit authorship contribution statement**

27 **M. Bonopera:** Formulation and evolution of overarching research goals and aims, Funding acquisition,
28 Methodology, Numerical validation of theory using commercial software, Substantial contribution to

1 conception, Writing - original draft. **K.-C. Chang**: Methodology, Substantial contribution to acquisition
2 of data results, Substantial contribution to replication of results. **N. Tullini**: Funding acquisition, Ideas,
3 Methodology, Substantial contribution to acquisition of data results, Substantial contribution to
4 conception, Writing - review & editing.

5 6 **Declaration of competing interest**

7 The authors declare that they have no known competing financial interests or personal relationships that
8 could have appeared to influence the work reported in this paper.

9 10 **Acknowledgments**

11 The tests were executed at the NCREE and funded by the National Applied Research Laboratories Project
12 of Taiwan [NCREE-06107A1700]. M.B. acknowledges the funding provided by the Ministry Of Science
13 and Technology (MOST) of Taiwan under the framework of the project “Recruitment of Visiting Science
14 and Technology Personnel” [MOST 109-2811-E-492-500]. N.T. acknowledges the financial support of
15 the “Research Program FAR 2022” provided by the University of Ferrara. Special gratitude is extended
16 to the laboratory technicians of the NCREE, who furnished assistance to the authors, and to Dr. Darragh
17 Noble of the University of Dublin for the help in providing several information about his experiments.

18 19 **Nomenclature**

20 A : cross sectional area of the thin-walled Rectangular Hollow Section (RHS)
21 A_f, A_0, A_1, A_2, A_3 : servo velocity seismometers
22 A_{tendon} : cross sectional area of straight tendon
23 A_w : cross sectional area of the web of a thin-walled RHS
24 b : internal width of a thin-walled RHS
25 e : eccentricity of straight tendon
26 E : Young’s modulus of the thin-walled member
27 E_{tendon} : Young’s modulus of straight tendon
28 $f_{I,I,FE,P\&S}$: first-order fundamental frequency based on FE plate/shell modeling
29 $f_{I,I,FE,P\&S,D}$: first-order fundamental frequency with a deviator based on FE plate/shell modeling
30 $f_{I,II,FE,P\&S}$: second-order fundamental frequency based on FE plate/shell modeling
31 $f_{I,II,FE,P\&S,1Link}$: second-order fundamental frequency with 1 rigid link based on FE plate/shell modeling
32 $f_{I,II,FE,P\&S,3Links}$: second-order fundamental frequency with 3 rigid links based on FE plate/shell modeling
33 $f_{I,II,FE,P\&S,D}$: second-order fundamental frequency with a deviator based on FE plate/shell modeling
34 $f_{I,II,FE,P\&S,NoMasses}$: second-order fundamental frequency without masses at the RHS ends based on FE
35 plate/shell modeling
36 $f_{1,exp}$: experimental fundamental frequency
37 $f_{1,exp,max}$: experimental maximum fundamental frequency
38 $f_{1,exp,min}$: experimental minimum fundamental frequency
39 $f_{E.-B.,I,I}$: first-order fundamental frequency based on Euler-Bernoulli model

1 $f_{E-B,1,I,FE}$: first-order fundamental frequency based on FE model
2 $f_{E-B,1,I,FE,W}$: first-order fundamental frequency with mass W based on FE model
3 $f_{E-B,1,II}$: second-order fundamental frequency based on Euler-Bernoulli model
4 $f_{E-B,1,II,FE}$: second-order fundamental frequency based on FE model
5 $f_{E-B,1,II,FE,W}$: second-order fundamental frequency with mass W based on FE model
6 $f_{E-B,n,II}$: second-order n th natural frequency based on Euler-Bernoulli model
7 $f_{shear,1,I}$: first-order fundamental frequency based on shear model
8 $f_{shear,1,I,FE}$: first-order fundamental frequency based on FE model including shear deformation
9 $f_{shear,1,I,FE,W}$: first-order fundamental frequency with mass W based on FE model including shear
10 deformation
11 $f_{shear,1,I,FE,W,D}$: first-order fundamental frequency with mass W and deviator based on FE model including
12 shear deformation
13 $f_{shear,1,II}$: second-order fundamental frequency based on shear model
14 $f_{shear,1,II,FE}$: second-order fundamental frequency based on FE model including shear deformation
15 $f_{shear,1,II,FE,W}$: second-order fundamental frequency with mass W based on FE model including shear
16 deformation
17 $f_{shear,1,II,FE,W,D}$: second-order fundamental frequency with mass W and deviator based on FE model
18 including shear deformation
19 $f_{shear,n,I}$: first-order n th natural frequency based on shear model
20 $f_{shear,n,II}$: second-order n th natural frequency based on shear model
21 F_d : release force of transverse vibration tests
22 f_{yk} : yielding stress of the thin-walled member
23 g : gravitational acceleration
24 G : shear modulus of the thin-walled member
25 h : height of a thin-walled RHS
26 i : measurement cross section along the thin-walled member
27 I : cross sectional second moment of the area of the thin-walled member
28 k : shear coefficient for a generic thin-walled RHS
29 L : length of the thin-walled member
30 m : coefficient considering the shear stresses throughout the cross sectional area of a thin-walled RHS
31 m_{beam} : thin-walled member's self-mass per unit length
32 m_{tendon} : straight tendon's self-mass per unit length
33 m_{tot} : thin-walled member's total self-mass per unit length corresponding to $m_{beam} + W/L$.
34 $m_{tot,beam}$: thin-walled member's total self-mass
35 M_μ : bending moments at the supports caused by sliding friction between steel-on-steel
36 n : vibration beam mode
37 N_{0x} : mean post-tensioning force
38 N_{0x1} : applied post-tensioning force at the left RHS end
39 N_{0x2} : applied post-tensioning force at the right RHS end
40 $N_{0x,aver}$: mean applied post-tensioning force
41 $N_{0x,aver,max}$: maximum mean applied post-tensioning force
42 $N_{0x,pl,0.12}$: mean post-tensioning force causing the plasticity of cross section when $e/h = 0.12$
43 $N_{0x,pl,0.23}$: mean post-tensioning force causing the plasticity of cross section when $e/h = 0.23$
44 $N_{0x,pl,0.28}$: mean post-tensioning force causing the plasticity of cross section when $e/h = 0.28$
45 $N_{0x,pl,0.74}$: mean post-tensioning force causing the plasticity of cross section when $e/h = 0.74$
46 $N_{0x,ULS,0.12}$: mean post-tensioning force causing the yielding stress of the most extreme fibres of cross
47 section when $e/h = 0.12$
48 $N_{0x,ULS,0.23}$: mean post-tensioning force causing the yielding stress of the most extreme fibres of cross
49 section when $e/h = 0.23$
50 $N_{0x,ULS,0.28}$: mean post-tensioning force causing the yielding stress of the most extreme fibres of cross
51 section when $e/h = 0.28$
52 $N_{0x,ULS,0.74}$: mean post-tensioning force causing the yielding stress of the most extreme fibres of cross
53 section when $e/h = 0.74$
54 $N_{crE,1}$: Euler buckling load

1 $N_{crE,n}$: n th critical buckling load
2 $N_{crE,hear,1}$: first critical buckling load including shear deformation
3 $N_{crE,hear,n}$: n th critical buckling load including shear deformation
4 r : radius of inertia of the thin-walled RHS
5 t : thickness of the web of a thin-walled RHS
6 t_f : thickness of the flanges of a thin-walled RHS
7 ν : Poisson's ratio of the thin-walled member
8 $v^{(0)}$: flexural shape of the fundamental frequency
9 W : mass of hydraulic oil jack
10 $\mu_{sliding}$: frictional coefficient regarding the contact between steel-on-steel surfaces at the supports
11 ρ_{beam} : unit weight of the thin-walled member
12 ρ_{tendon} : unit weight of straight tendon
13 $\sigma_{tendon,max}$: maximum tensile strength reached in the straight tendon
14 σ_{uy} : yielding stress of straight tendon
15
16

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